



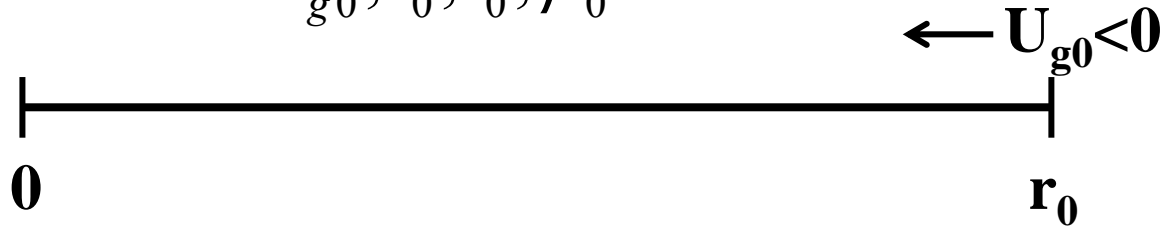
# CONVERGENT SHOCK IN A GAS FOR LARGE VALUES OF A SELF-SIMILAR COEFFICIENT

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In formulating of the problem we assign four parameters with different dimensions:  $U_{g0}, r_0, t_0, \rho_0$ .



Initial parameters for the gas:  $\rho_0 = const, U_0 = 0, P_0 = 0, E_0 = 0$

Conservation laws on the shock wave:

$$\rho_w (D - U_w) - \rho_0 D = 0, \quad (1)$$

$$\rho_0 D U_w - P_w = 0, \quad (2)$$

$$\rho_0 D \left( E_w + \frac{1}{2} U_w^2 \right) - P_w U_w = 0. \quad (3)$$

Equations (1) - (3) are closed by the equation of state

$$P = (\gamma - 1) \rho E, \quad P = F(S) \rho^\gamma. \quad (4)$$

$M_w = \frac{4}{3} \pi \rho_0 r_w^3$  - the Lagrangian coordinate of the shock wave

$W = (3M_w)^{2/3} (4\pi\rho_0)^{1/3} D$  - velocity of the shock wave  
in Lagrangian coordinates

Expressed  $D$  and substituting in (1) - (3), we obtain the conditions on the shock wave, which contain  $W$  and  $M_w$

$$\left( \frac{1}{\rho_w} - \frac{1}{\rho_0} \right) W + (4\pi)^{1/3} \left( \frac{3M_w}{\rho_0} \right)^{2/3} U_w = 0, \quad (5)$$

$$U_w W - (4\pi)^{1/3} \left( \frac{3M_w}{\rho_0} \right)^{2/3} P_w = 0, \quad (6)$$

$$\left( E_w + 0,5U_w^2 \right) W - (4\pi)^{1/3} \left( \frac{3M_w}{\rho_0} \right)^{2/3} P_w U_w = 0. \quad (7)$$

The expressions follow from (4), (5), (6) and (7)

$$\rho_w = \frac{\gamma + 1}{\gamma - 1} \rho_0,$$

$$U_w = \frac{2W}{(\gamma + 1)(4\pi\rho_0)^{1/3} (3M_w)^{2/3}},$$

$$P_w = \frac{2\rho_0^{1/3} W^2}{(\gamma + 1)(4\pi)^{2/3} (3M_w)^{4/3}},$$

$$F_w = \frac{2}{\gamma + 1} \left( \frac{\gamma - 1}{\gamma + 1} \right)^\gamma \frac{W^2 \rho_0^{(1-3\gamma)/3}}{(4\pi)^{2/3} (3M_w)^{4/3}}.$$

We define the trajectory of the shock wave in the form

$$M_w = M_0 \left( \frac{t_f - t}{t_f - t_0} \right)^n, \quad (8)$$

where  $t_f$  – focusing time.

$$W = W_0 \left( \frac{t_f - t}{t_f - t_0} \right)^{n-1},$$

where  $W_0 = 3M_0^{2/3} (4\pi\rho_0)^{1/3} \frac{(\gamma + 1)}{2} U_g$ ,

$$t_f = t_0 - \frac{M_0 n}{W_0}.$$

Parameters of the adiabatic flow between the shock wave and boundary of the gas are determined by the equations of the trajectory, the conservation of mass and motion

$$\left(\frac{\partial r}{\partial t}\right)_M - U = 0, \quad (9)$$

$$\left(\frac{\partial \rho}{\partial t}\right)_M + 4\pi\rho^2 \frac{\partial(r^2 U)}{\partial M} = 0, \quad (10)$$

$$\left(\frac{\partial U}{\partial t}\right)_M + 4\pi r^2 \frac{\partial(F\rho^\gamma)}{\partial M} = 0. \quad (11)$$

Let us pass in (9) - (11) to new desired functions

$$R = r^3, \quad C = r^2 U.$$

After the pass the equations (9) - (11) take the form

$$\left( \frac{\partial R}{\partial t} \right)_M - 3C = 0, \tag{12}$$

$$\left( \frac{\partial \rho}{\partial t} \right)_M + 4\pi\rho^2 \frac{\partial C}{\partial M} = 0, \tag{13}$$

$$\left( \frac{\partial C}{\partial t} \right)_M + 4\pi R^3 \frac{\partial (F\rho^\gamma)}{\partial M} - 2C^2 R^{-1} = 0. \tag{14}$$

New functions on the shock wave have the form

$$R_w = R_0 \frac{M_w}{M_0}, \quad C_w = C_0 \left( \frac{M_w}{M_0} \right)^{(n-1)/n} \tag{15}$$

Let us proceed from the variables  $t, M$  to variables  $t, \xi(t, M)$ .  
 We define the dependence  $\xi(t, M)$  in the form

$$\xi = \frac{M}{M_0} \left( \frac{t_f - t}{t_f - t_0} \right)^{-n}. \quad (16)$$

With the help of the equations for the derivatives

$$\left( \frac{\partial}{\partial t} \right)_M = \left( \frac{\partial}{\partial t} \right)_\xi + \left( \frac{\partial}{\partial \xi} \right)_t \left( \frac{\partial \xi}{\partial t} \right)_M, \quad \left( \frac{\partial}{\partial M} \right)_t = \left( \frac{\partial}{\partial \xi} \right)_t \left( \frac{\partial \xi}{\partial M} \right)_t$$

where

$$\frac{\partial \xi}{\partial t} = \frac{n\xi}{t_f - t}, \quad \frac{\partial \xi}{\partial M} = \frac{1}{M_0} \left( \frac{t_f - t}{t_f - t_0} \right)^{-n},$$



we transform the equations (12)–(14)

$$\left(\frac{\partial R}{\partial t}\right)_{\xi} + \left(\frac{\partial R}{\partial \xi}\right)_t \left(\frac{\partial \xi}{\partial t}\right)_M - 3C = 0, \quad (17)$$

$$\left(\frac{\partial \rho}{\partial t}\right)_{\xi} + \left(\frac{\partial \rho}{\partial \xi}\right)_t \left(\frac{\partial \xi}{\partial t}\right)_M + 4\pi\rho^2 \left(\frac{\partial C}{\partial \xi}\right)_t \left(\frac{\partial \xi}{\partial M}\right)_t = 0, \quad (18)$$

$$\left(\frac{\partial C}{\partial t}\right)_{\xi} + \left(\frac{\partial C}{\partial \xi}\right)_t \left(\frac{\partial \xi}{\partial t}\right)_M - \frac{2C^2}{R} + \quad (19)$$

$$+ 4\pi R^{\frac{4}{3}} \left[ \rho^{\gamma} \left(\frac{\partial F}{\partial \xi}\right)_t \left(\frac{\partial \xi}{\partial M}\right)_t + \gamma F \rho^{\gamma-1} \left(\frac{\partial \rho}{\partial \xi}\right)_t \left(\frac{\partial \xi}{\partial M}\right)_t \right] = 0.$$

To separate the variables representing  $R$ ,  $\rho$  and  $C$  in the form of products of functions of the time and the function of  $\xi$

$$R = \alpha_R(t)T(\xi) \quad \rho = \alpha_\rho(t)\delta(\xi) \quad C = \alpha_C(t)Z(\xi)$$

We obtain these dependencies on the shock wave at  $\xi=1$  with the help (14) and (15)

$$R_w = R_0 \cdot \left( \frac{t_f - t}{t_f - t_0} \right)^n, \quad C_w = C_0 \cdot \left( \frac{t_f - t}{t_f - t_0} \right)^{n-1}. \quad (20)$$

Then

$$T_w = T_1, \quad \alpha_R(t) = R_0 \left( \frac{t_f - t}{t_f - t_0} \right)^n T_1^{-1},$$

$$\delta_w = \delta_1, \quad \alpha_\rho = \rho_0 \left( \frac{\gamma + 1}{\gamma - 1} \right) \delta_1^{-1}, \quad (21)$$

$$Z_w = Z_1, \quad \alpha_C(t) = C_0 \left( \frac{t_f - t}{t_f - t_0} \right)^{n-1} Z_1^{-1}.$$

By substituting (20), (21) in (17)–(19), we obtain three equations for  $T$ ,  $\delta$  and  $Z$

$$\xi T' = A_1, \quad (22)$$

$$\delta_1 B_1 Z' - \xi Z_1 \delta' = 0, \quad (23)$$

$$-\xi Z_1^{-1} Z' + C_1 \gamma \xi \delta_1^{-1} \delta' = C_2, \quad (24)$$

where

$$A_1 = T - \frac{2ZT_1}{(\gamma+1)Z_1}, \quad B_1 = \frac{2\delta^2}{(\gamma-1)\delta_1^2}, \quad C_1 = \frac{T^{4/3}\delta^{\gamma-1}\xi^{-(n+6)/3n}}{T_1^{4/3}\delta_1^{\gamma-1}},$$

$$C_2 = \frac{4Z^2T_1}{3(\gamma+1)Z_1^2T} - \frac{(n-1)Z}{nZ_1} - C_1 \frac{2(n-3)\delta}{3n\delta_1}.$$

The equations (22) - (24) are the system of linear inhomogeneous equations regarding to the  $T'$ ,  $\delta'$ ,  $Z'$ .

The determinant of the system is the following

$$\Delta = B_1 C_1 \gamma \xi - \xi^2.$$

If  $\Delta \neq 0$ , the solution of the system (22)–(24) has the form

$$T' = \frac{A_1}{\xi}, \quad \delta' = \frac{B_1 C_2}{\Delta}, \quad Z' = \frac{\xi C_2}{\Delta}.$$

The values  $n_*$  and  $\xi_*$ , in which simultaneously  $\Delta(\xi_*)=0$  and  $C_2(\xi_*)=0$  corresponding to the values  $\gamma$  are given in the table

$\gamma$	1.1	1.2	4/3	1.4	5/3
$n_*$	2.387916	2.271434	2.183068	2.151532	2.065135
$\xi_*$	7.959997	5.717071	4.559431	4.227062	3.481885

In the area  $n < n_*$   $\Delta(\xi) > 0$  for  $1 \leq \xi < \infty$

In the area  $n > n_*$   $\Delta(\xi_n) = 0$ , but  $C_2(\xi_n) \neq 0$ . On the border of the gas sphere at  $M = M_0$  the value of  $\xi_n$  is reached at the moment

$$t_n = t_f - (t_f - t_0) \xi_n^{-1/n}.$$

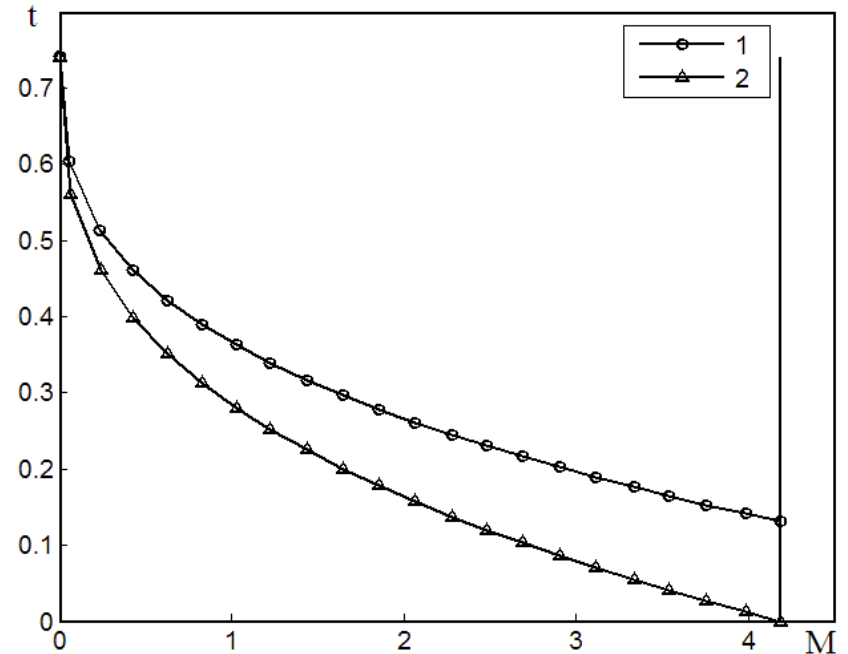
From the point  $M_0, t_n$  comes out the line, on which  $\xi = \xi_n$ .

The equation of line is

$$M = M_0 \xi_n \left( \frac{t_f - t}{t_f - t_0} \right)^n. \quad (25)$$

This is a characteristic. This line is focuses together with the shock wave, because  $M=0$  at  $t=t_f$ . In the area between (25) and the shock wave (8) for each  $n > n_*$  the single solution is exists.

In Fig. 1 – It's a shock wave,  
2 – It's a characteristic.



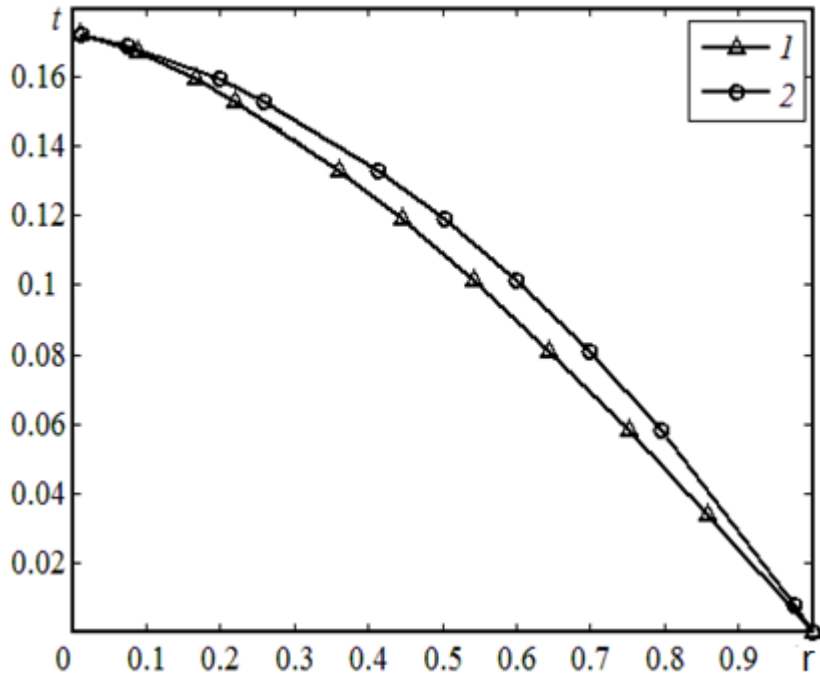


Fig. 1 The trajectories of the shock wave and the border of the gas sphere for  $\gamma=5/3$  and  $n=0.68$ .

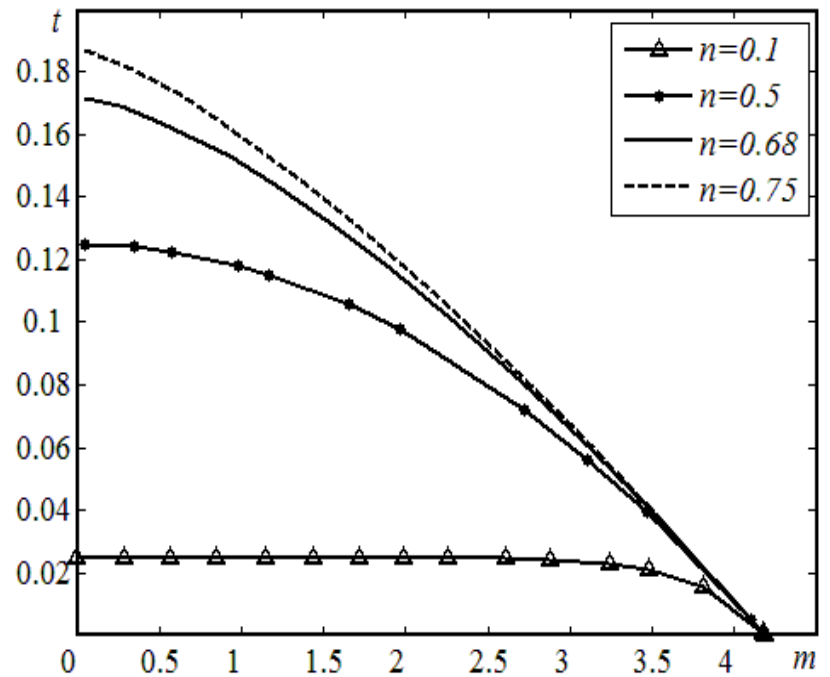


Fig. 2 The trajectories of the shock wave for  $\gamma=5/3$  and four values  $n=0.1; 0.5; 0.68; 0.75$ .

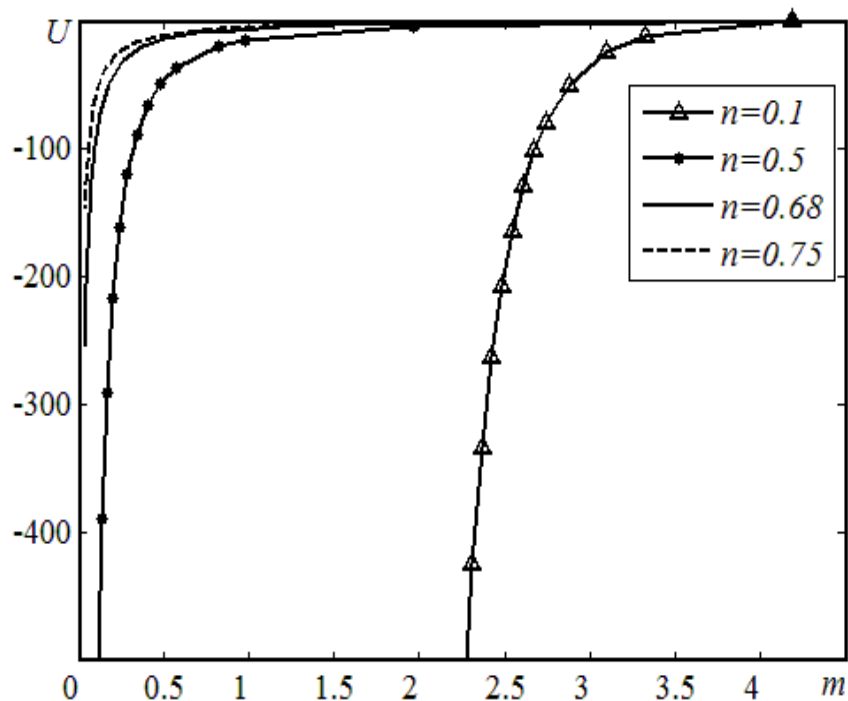


Fig. 3 Profiles of velocity on the shock wave for  $\gamma=5/3$  and four values  $n=0.1; 0.5; 0.68; 0.75$  in Lagrangian coordinates.

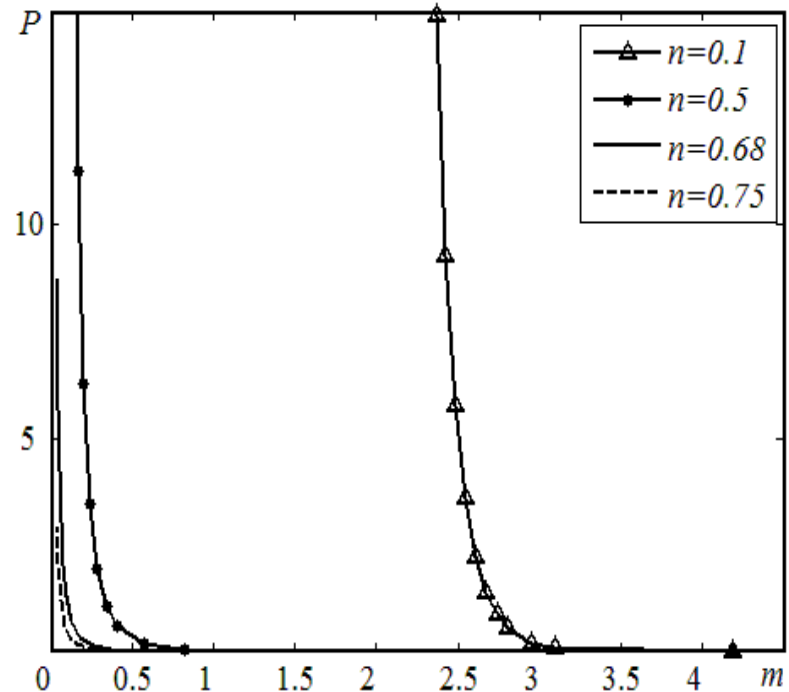


Fig. 4 Profiles of pressure on the shock wave for  $\gamma=5/3$  and four values  $n=0.1; 0.5; 0.68; 0.75$  in Lagrangian coordinates.

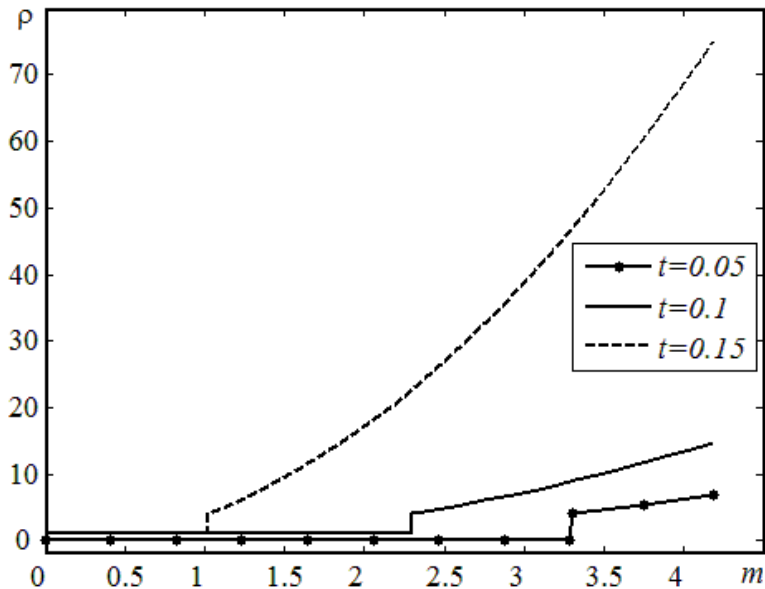
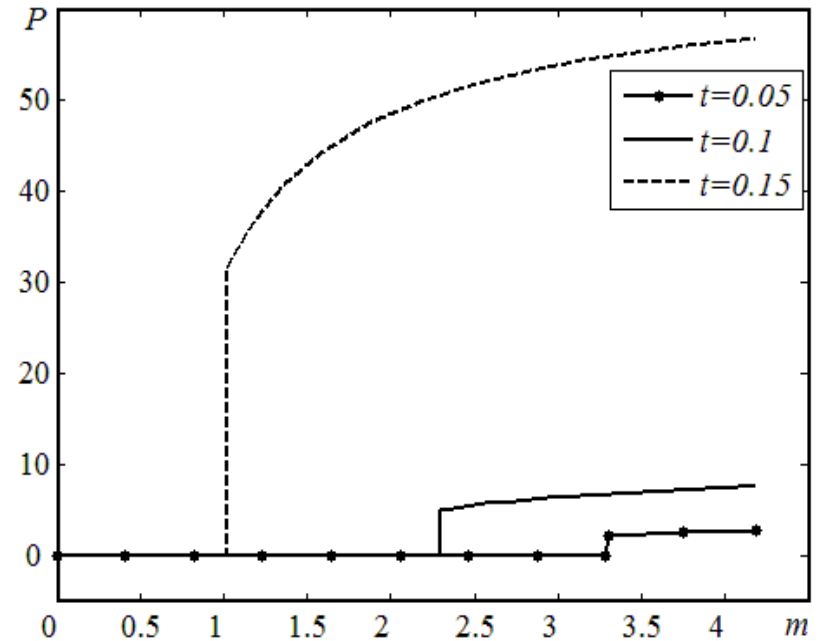
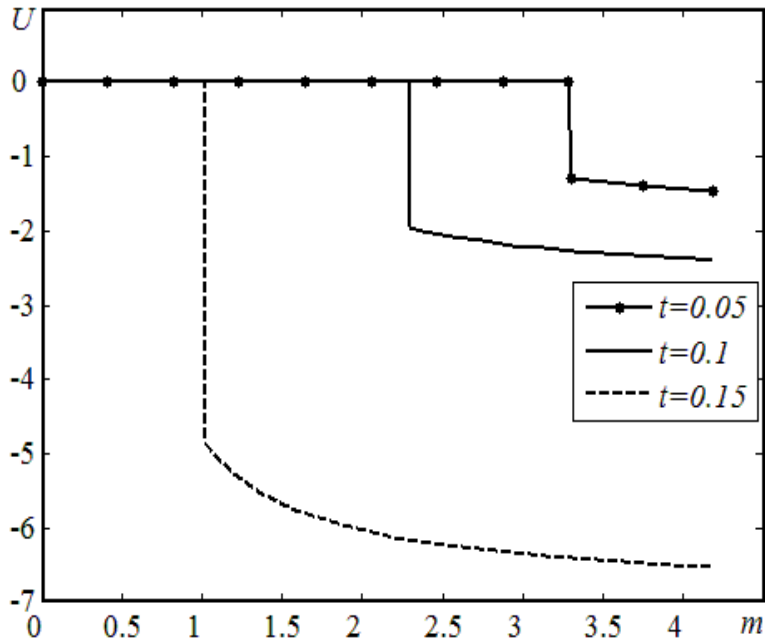


Fig. 5 Profiles of velocity, pressure and density for  $\gamma=5/3$  and  $n=0.68$  for time points  $t=0.05; 0.1; 0.15$  in Lagrangian coordinates.



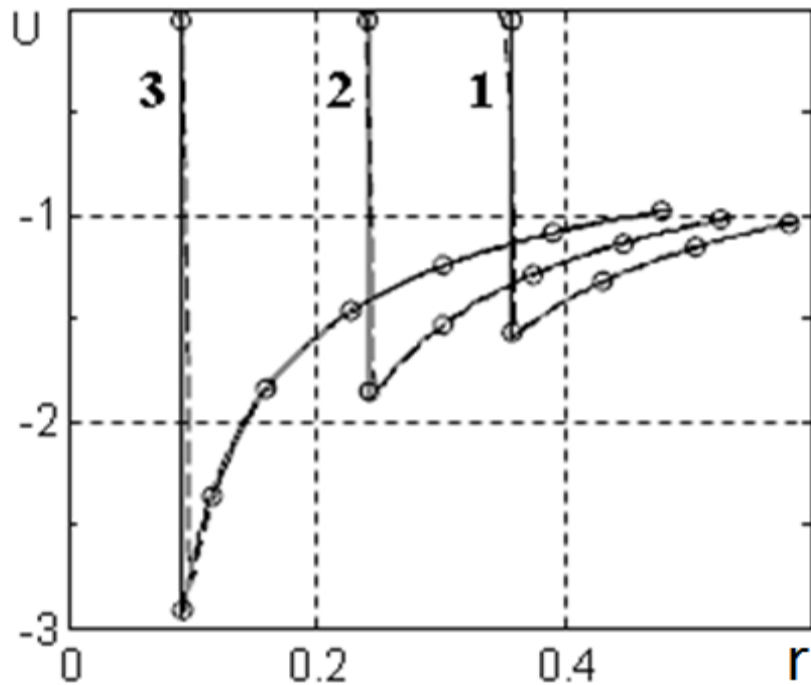
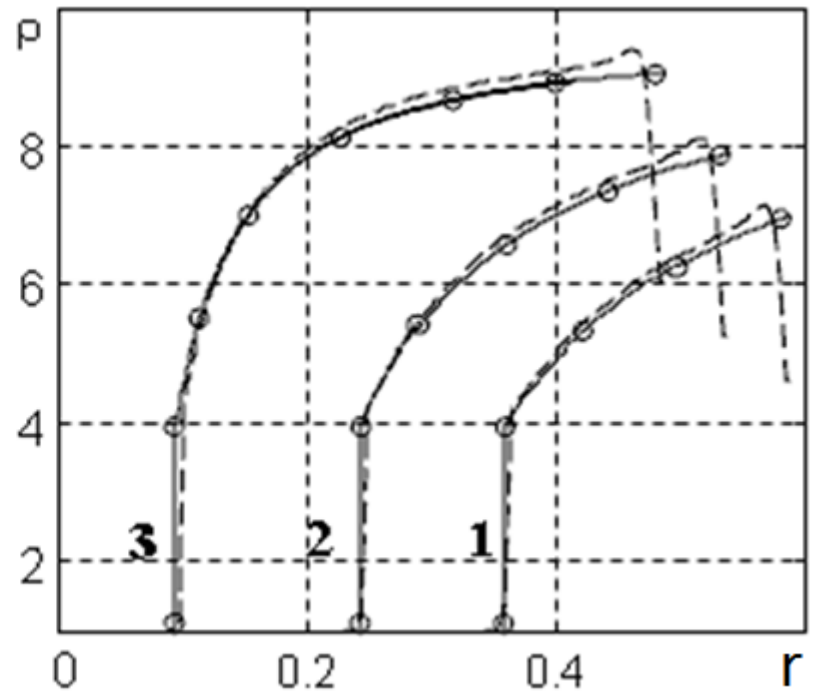
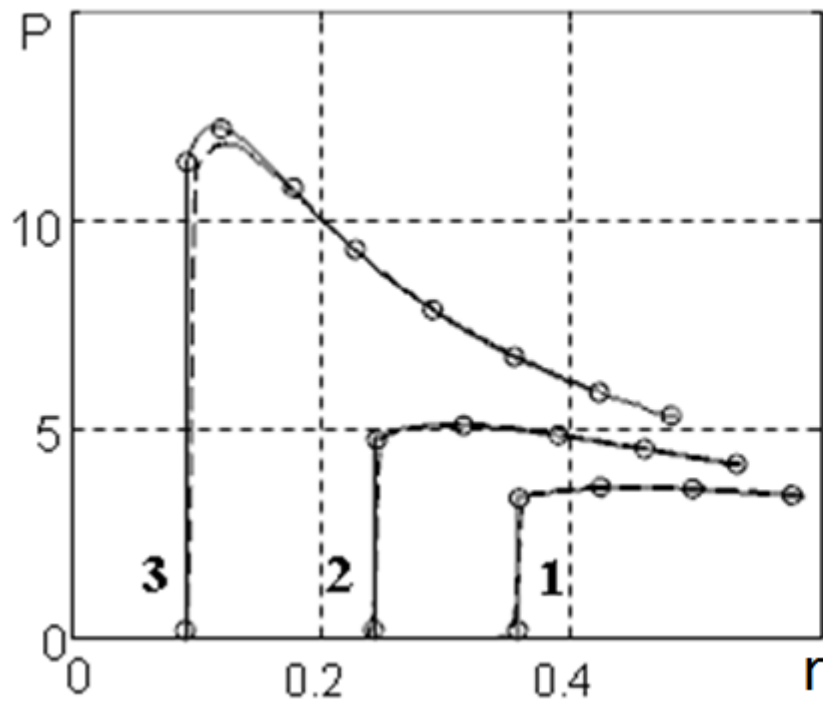


Fig. 6 Profiles of pressure, density and velocity for  $\gamma=5/3$  and  $n=n_*=2,065135$  for time points 1- $t=0.4$ ; 2- $t=0.45$ ; 3- $t=0.5$  in Euler coordinates.

The solid line – it's the analytical solution of this work,

-o- – the calculations on VOLNA program with the selection of discontinuity,

--- – calculations on VOLNA program without the selection of discontinuity.

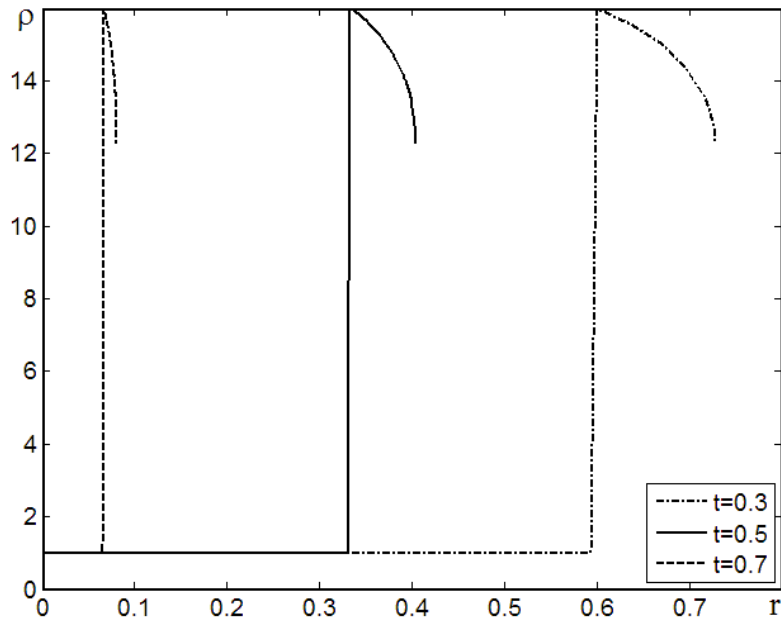
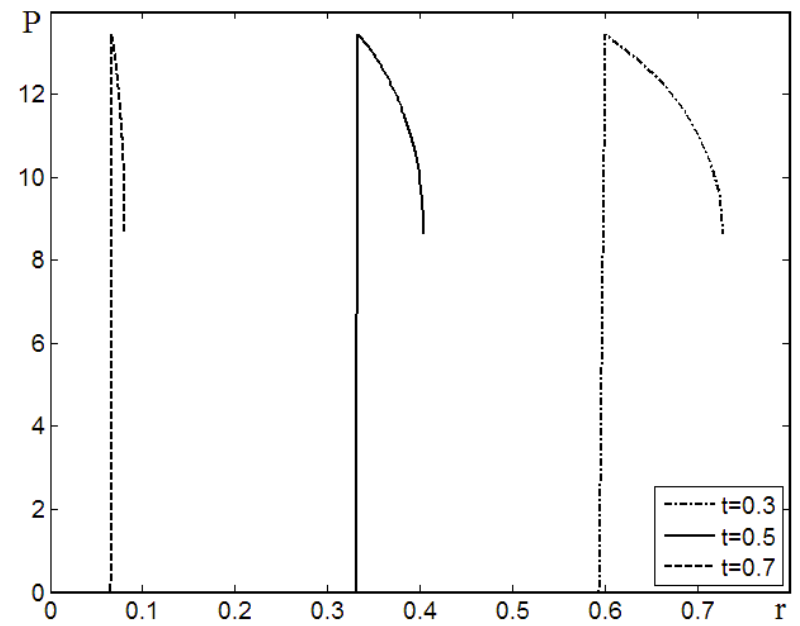
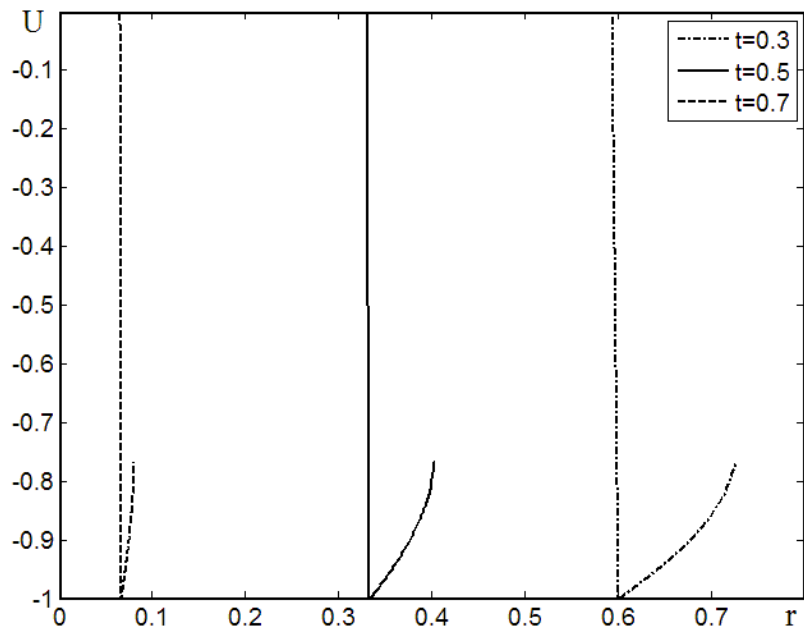


Fig. 7 Profiles of velocity, pressure and density between the shock wave and the characteristic for  $\gamma=5/3$  and  $n=3$  for time points  $t=0.3; 0.5; 0.7$  in Euler coordinates.

Thank you for attention!